



STIC Search Report

Biotech-Chem Library

STIC Database Tracking Number: 203077

TO: Aaron Austin
Location: Remsen 5e74
Wednesday, September 27, 2006
Art Unit: 1775
Phone: 571-272-8935
Serial Number: 10 / 792003

From: Jan Delaval
Location: EIC 1700
Remsen 4a30
Phone: 571-272-2504

jan.delaval@uspto.gov

Search Notes

For claims 10, 12 & 13 → WO ref → corresponds to present
→ cl 16 was not found ^{appl}
filed on same date

Banks, Kendra

203077

From: AARON AUSTIN [Aaron.Austin@uspto.gov]
Sent: Wednesday, September 27, 2006 8:28 AM
To: STIC-EIC1700
Subject: Database Search Request, Serial Number: 10792003

Requester:
AARON AUSTIN (P/1775)
Art Unit:
GROUP ART UNIT 1775
Employee Number:
82019
Office Location:
REM 05E74
Phone Number:
(571)272-8935
Mailbox Number:

SCIENTIFIC REFERENCE BR
Sci & Tech Inf. Cntr

SEP 27 RECD

Pat. & T.M. Office

Case serial number:
10792003
Class / Subclass(es):
428/678,636,637,926,934; 416/241R
Earliest Priority Filing Date:
3/2/04
Format preferred for results:
E-mail

Search Topic Information:

I am looking for nickel-based alloys (particularly MCrAlY bond coats for turbine engines) with the following 4 embodiments:

Claim 10

Element - Range Weight %
Co - about 15 - about 22
Cr - about 15- about 25
Al - about 8- about 15
Y - about 0.1- about 1.0
Pt - about 20- about 35
Hf - about 1.0- about 5.0
Si - about 1.0- about 5.0
Zr - 0 - about 3.0
Ta - 0 - about 5.0
Re - about 1.0- about 5.0
Ru - about 1.0- about 5.0
Ni - remainder

Claim 12

Element - Weight %
Co - about 20
Cr - about 25
Al - about 13
Y - about 0.3
Hf - about 2.0
Si - about 0.65
Re - about 3.0
Ni - remainder

Claim 13

Element - Weight %
Co - about 20
Cr - about 22

Al - about 13
Y - about 0.3
Hf - about 2.0
Si - about 0.65
Re - about 3.0
Ru - about 1.5
Ni - remainder

Claim 16

Element - Weight %

Co - about 15 - about 22
Cr - about 15 - about 25
Al - about 8 - about 15
Y - about 0.1 - about 1.0
Hf - about 1.0 - about 5.0
Si - about 1.0 - about 5.0
Zr - about 1.0 - about 3.0
Ta - about 1.0 - about 5.0
Re - about 1.0 - about 5.0
Ru - about 1.0 - about 5.0
N - remainder

Special Instructions and Other Comments:



STIC Search Results Feedback Form

EIC17000

Questions about the scope or the results of the search? Contact the EIC searcher or contact:

Kathleen Fuller, EIC 1700 Team Leader
571/272-2505 REMSEN 4B28

Voluntary Results Feedback Form

- I am an examiner in Workgroup: Example: 1713
➤ Relevant prior art found, search results used as follows.

- ☐ 102 rejection
- ☐ 103 rejection
- ☐ Cited as being of interest.
- ☐ Helped examiner better understand the invention.
- ☐ Helped examiner better understand the state of the art in their technology.

Types of relevant prior art found:

- ☐ Foreign Patent(s)
- ☐ Non-Patent Literature
(journal articles, conference proceedings, new product announcements etc.)

➤ Relevant prior art not found:

- ☐ Results verified the lack of relevant prior art (helped determine patentability).
- ☐ Results were not useful in determining patentability or understanding the invention.

Comments:

=> fil reg

FILE 'REGISTRY' ENTERED AT 16:05:17 ON 27 SEP 2006
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provided by InfoChem.

STRUCTURE FILE UPDATES: 26 SEP 2006 HIGHEST RN 908803-03-2
DICTIONARY FILE UPDATES: 26 SEP 2006 HIGHEST RN 908803-03-2

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TSCA INFORMATION NOW CURRENT THROUGH June 30, 2006

Please note that search-term pricing does apply when
conducting SmartSELECT searches.

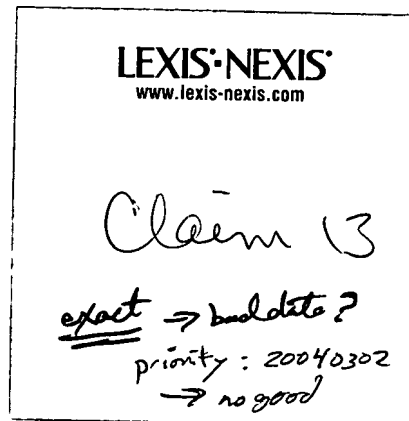
REGISTRY includes numerically searchable data for experimental and
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experimental property data in the original document. For information
on property searching in REGISTRY, refer to:

<http://www.cas.org/ONLINE/UG/regprops.html>

=> d l42 ide can tot

L42 ANSWER 1 OF 6 REGISTRY COPYRIGHT 2006 ACS on STN
RN 880354-78-9 REGISTRY
ED Entered STN: 13 Apr 2006
CN Nickel alloy, base, Ni 38,Cr 22,Co 20,Al 13,Re 3,Hf 2,Ru 1.5,Si 0.6,Y 0.3
(9CI) (CA INDEX NAME)
MF Al . Co . Cr . Hf . Ni . Re . Ru . Si . Y
CI AYS
SR CA
LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Ni	38	7440-02-0
Cr	22	7440-47-3
Co	20	7440-48-4
Al	13	7429-90-5
Re	3	7440-15-5
Hf	2	7440-58-6
Ru	1.5	7440-18-8
Si	0.6	7440-21-3
Y	0.3	7440-65-5



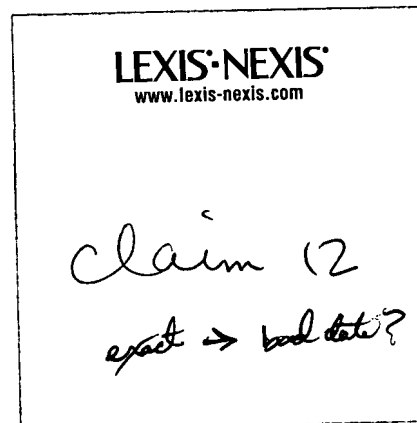
1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

REFERENCE 1: 144:335551

L42 ANSWER 2 OF 6 REGISTRY COPYRIGHT 2006 ACS on STN
RN 880354-77-8 REGISTRY
ED Entered STN: 13 Apr 2006
CN Nickel alloy, base, Ni 36,Cr 25,Co 20,Al 13,Re 3,Hf 2,Si 0.6,Y 0.3 (9CI)

(CA INDEX NAME)
 MF Al . Co . Cr . Hf . Ni . Re . Si . Y
 CI AYS
 SR CA
 LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Ni	36	7440-02-0
Cr	25	7440-47-3
Co	20	7440-48-4
Al	13	7429-90-5
Re	3	7440-15-5
Hf	2	7440-58-6
Si	0.6	7440-21-3
Y	0.3	7440-65-5

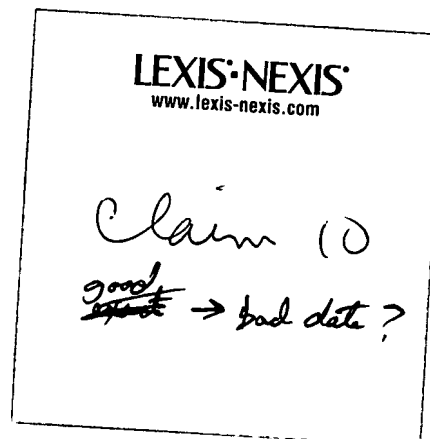


1 REFERENCES IN FILE CA (1907 TO DATE)
 1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

REFERENCE 1: 144:335551

L42 ANSWER 3 OF 6 REGISTRY COPYRIGHT 2006 ACS on STN
 RN 880354-76-7 REGISTRY
 ED Entered STN: 13 Apr 2006
 CN Nickel alloy, base, Ni 0-62, Pt 0-35, Cr 15-25, Co 15-22, Al 8-15, Hf 0-5, Re 0-5, Ru 0-5, Si 0-5, Ta 0-5, Zr 0-3, Y 0.1-1 (9CI) (CA INDEX NAME)
 MF Al . Co . Cr . Hf . Ni . Pt . Re . Ru . Si . Ta . Y . Zr
 CI AYS
 SR CA
 LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Ni	0 - 62	7440-02-0
Pt	0 - 35	7440-06-4
Cr	15 - 25	7440-47-3
Co	15 - 22	7440-48-4
Al	8 - 15	7429-90-5
Hf	0 - 5	7440-58-6
Re	0 - 5	7440-15-5
Ru	0 - 5	7440-18-8
Si	0 - 5	7440-21-3
Ta	0 - 5	7440-25-7
Zr	0 - 3	7440-67-7
Y	0.1 - 1	7440-65-5



1 REFERENCES IN FILE CA (1907 TO DATE)
 1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

REFERENCE 1: 144:335551

L42 ANSWER 4 OF 6 REGISTRY COPYRIGHT 2006 ACS on STN
 RN 880354-75-6 REGISTRY
 ED Entered STN: 13 Apr 2006
 CN Nickel alloy, base, Ni 0-36, Pt 20-35, Cr 15-25, Co 15-22, Al 8-15, Hf 1-5, Re 1-5, Ru 1-5, Si 1-5, Ta 1-5, Zr 1-3, Y 0.1-1 (9CI) (CA INDEX NAME)
 MF Al . Co . Cr . Hf . Ni . Pt . Re . Ru . Si . Ta . Y . Zr

CI AYS
SR CA
LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Ni	0 - 36	7440-02-0
Pt	20 - 35	7440-06-4
Cr	15 - 25	7440-47-3
Co	15 - 22	7440-48-4
Al	8 - 15	7429-90-5
Hf	1 - 5	7440-58-6
Re	1 - 5	7440-15-5
Ru	1 - 5	7440-18-8
Si	1 - 5	7440-21-3
Ta	1 - 5	7440-25-7
Zr	1 - 3	7440-67-7
Y	0.1 - 1	7440-65-5

LEXIS-NEXIS
www.lexis-nexis.com

Claim 10
good (almost exact)
→ bad date?

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

REFERENCE 1: 144:335551

L42 ANSWER 5 OF 6 REGISTRY COPYRIGHT 2006 ACS on STN
RN 540788-35-0 REGISTRY
ED Entered STN: 02 Jul 2003
CN Nickel alloy, base, Ni 61, Cr 18, Co 10, Al 6, Re 2, Si 1.6, Y 1, Hf 0.8 (9CI)
(CA INDEX NAME)
MF Al . Co . Cr . Hf . Ni . Re . Si . Y
CI AYS
SR CA
LC STN Files: CA, CAPLUS, USPAT2, USPATFULL

Component	Component Percent	Component Registry Number
Ni	61	7440-02-0
Cr	18	7440-47-3
Co	10	7440-48-4
Al	6	7429-90-5
Re	2	7440-15-5
Si	1.6	7440-21-3
Y	1	7440-65-5
Hf	0.8	7440-58-6

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

LEXIS-NEXIS
www.lexis-nexis.com

Claim 12
not close enough

REFERENCE 1: 139:40128

L42 ANSWER 6 OF 6 REGISTRY COPYRIGHT 2006 ACS on STN
RN 540788-32-7 REGISTRY
ED Entered STN: 02 Jul 2003
CN Nickel alloy, base, Ni 62, Cr 18, Co 10, Al 6.5, Re 2, Si 1, Hf 0.5, Y 0.3 (9CI)
(CA INDEX NAME)
MF Al . Co . Cr . Hf . Ni . Re . Si . Y
CI AYS
SR CA

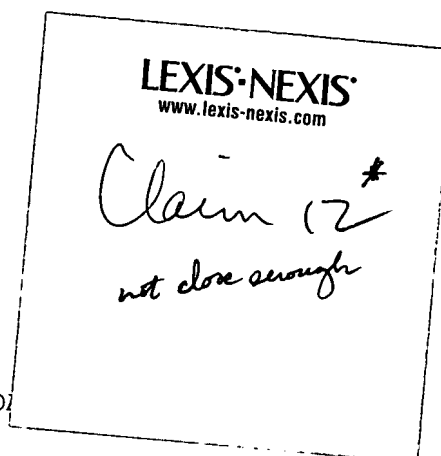
LC STN Files: CA, CAPLUS, USPAT2, USPATFULL

Component	Component Percent	Component Registry Number
Ni	62	7440-02-0
Cr	18	7440-47-3
Co	10	7440-48-4
Al	6.5	7429-90-5
Re	2	7440-15-5
Si	1	7440-21-3
Hf	0.5	7440-58-6
Y	0.3	7440-65-5

1 REFERENCES IN FILE CA (1907 TO DATE)

1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

REFERENCE 1: 139:40128



=> fil uspatful

FILE 'USPATFULL' ENTERED AT 16:05:24 ON 27 SEP 2006

CA INDEXING COPYRIGHT (C) 2006 AMERICAN CHEMICAL SOCIETY (ACS)

FILE COVERS 1971 TO PATENT PUBLICATION DATE: 26 Sep 2006 (20060926/PD)

FILE LAST UPDATED: 26 Sep 2006 (20060926/ED)

HIGHEST GRANTED PATENT NUMBER: US7114185

HIGHEST APPLICATION PUBLICATION NUMBER: US2006212984

CA INDEXING IS CURRENT THROUGH 26 Sep 2006 (20060926/UPCA)

ISSUE CLASS FIELDS (/INCL) CURRENT THROUGH: 26 Sep 2006 (20060926/PD)

REVISED CLASS FIELDS (/NCL) LAST RELOADED: Jun 2006

USPTO MANUAL OF CLASSIFICATIONS THESAURUS ISSUE DATE: Jun 2006

=> d bib abs hitstr 147

L47 ANSWER 1 OF 1 USPATFULL on STN

AN 2003:213503 USPATFULL

TI High temperature corrosion resistant alloy, thermal barrier coating material, and gas turbine using high temperature corrosion resistant alloy

IN Oguma, Hidetaka, Takasago-shi, JAPAN

Okada, Ikuo, Takasago-shi, JAPAN

Torigoe, Taiji, Takasago-shi, JAPAN

Takahashi, Kouji, Takasago-shi, JAPAN

PA MITSUBISHI HEAVY INDUSTRIES, LTD., Tokyo, JAPAN (non-U.S. corporation)

PI US 2003148140 A1 20030807

US 6756131 B2 20040629

AI US 2002-316070 A1 20021211 (10)

PRAI JP 2001-383689 20011217

DT Utility

FS APPLICATION

LREP OBLON, SPIVAK, MCCLELLAND, MAIER & NEUSTADT, P.C., 1940 DUKE STREET, ALEXANDRIA, VA, 22314

CLMN Number of Claims: 10

ECL Exemplary Claim: 1

DRWN 5 Drawing Page(s)

LN.CNT 694

CAS INDEXING IS AVAILABLE FOR THIS PATENT.

AB A high temperature corrosion resistant alloy composition comprising, in

addition to Ni, 0.1 to 12% by weight of Co, 10 to 30% by weight of Cr, 4 to 15% by weight of Al, 0.1 to 5% by weight of Y, and 0.5 to 10% by weight of Re. The high temperature corrosion resistant alloy composition has an excellent oxidation resistance and ductility and is suitable for use in a bonding layer of a thermal barrier coating material.

CAS INDEXING IS AVAILABLE FOR THIS PATENT.

IT 540788-32-7 540788-35-0

(oxidation resistant and ductile alloy, bonding layer under ceramic thermal barrier coating; high-temperature corrosion-resistant alloy and ceramic thermal barrier coating material with metal bonding layer used in gas turbine)

RN 540788-32-7 USPATFULL

CN Nickel alloy, base, Ni 62,Cr 18,Co 10,Al 6.5,Re 2,Si 1,Hf 0.5,Y 0.3 (9CI)
(CA INDEX NAME)

Component	Component Percent	Component Registry Number
Ni	62	7440-02-0
Cr	18	7440-47-3
Co	10	7440-48-4
Al	6.5	7429-90-5
Re	2	7440-15-5
Si	1	7440-21-3
Hf	0.5	7440-58-6
Y	0.3	7440-65-5

RN 540788-35-0 USPATFULL

CN Nickel alloy, base, Ni 61,Cr 18,Co 10,Al 6,Re 2,Si 1.6,Y 1,Hf 0.8 (9CI)
(CA INDEX NAME)

Component	Component Percent	Component Registry Number
Ni	61	7440-02-0
Cr	18	7440-47-3
Co	10	7440-48-4
Al	6	7429-90-5
Re	2	7440-15-5
Si	1.6	7440-21-3
Y	1	7440-65-5
Hf	0.8	7440-58-6

=> fil hcaplus

FILE 'HCAPLUS' ENTERED AT 16:05:37 ON 27 SEP 2006

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FILE COVERS 1907 - 27 Sep 2006 VOL 145 ISS 14
FILE LAST UPDATED: 26 Sep 2006 (20060926/ED)

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This file contains CAS Registry Numbers for easy and accurate substance identification.

=> d 146 all hitstr tot

L46 ANSWER 1 OF 2 HCAPLUS COPYRIGHT 2006 ACS on STN
AN 2006:333261 HCAPLUS
DN 144:335551
ED Entered STN: 12 Apr 2006
TI Modified **MCrAlY** coatings on turbine blade tips with improved durability
IN **Hu, Yiping; Hehmann, William, F.**
PA **Honeywell International Inc., USA**
SO PCT Int. Appl., 36 pp.
CODEN: PIXXD2
DT Patent
LA English
IC ICM C22C
CC 56-6 (Nonferrous Metals and Alloys)
FAN.CNT 1

	PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
PI	WO 2006025865	A2	20060309	WO 2005-US6833	20050302 <--
	WO 2006025865	A3	20060615		
	W:	AE, AG, AL, AM, AT, AU, AZ, BA, BB, BG, BR, BW, BY, BZ, CA, CH, CN, CO, CR, CU, CZ, DE, DK, DM, DZ, EC, EE, EG, ES, FI, GB, GD, GE, GH, GM, HR, HU, ID, IL, IN, IS, JP, KE, KG, KP, KR, KZ, LC, LK, LR, LS, LT, LU, LV, MA, MD, MG, MK, MN, MW, MX, MZ, NA, NI, NO, NZ, OM, PG, PH, PL, PT, RO, RU, SC, SD, SE, SG, SK, SL, SM, SY, TJ, TM, TN, TR, TT, TZ, UA, UG, US, UZ, VC, VN, YU, ZA, ZM, ZW			
	RW:	AT, BE, BG, CH, CY, CZ, DE, DK, EE, ES, FI, FR, GB, GR, HU, IE, IS, IT, LT, LU, MC, NL, PL, PT, RO, SE, SI, SK, TR, BF, BJ, CF, CG, CI, CM, GA, GN, GQ, GW, ML, MR, NE, SN, TD, TG, BW, GH, GM, KE, LS, MW, MZ, NA, SD, SL, SZ, TZ, UG, ZM, ZW, AM, AZ, BY, KG, KZ, MD, RU, TJ, TM			
PRAI	US 2004-792003	A	20040302	<--	

CLASS

PATENT NO.	CLASS	PATENT FAMILY CLASSIFICATION CODES
WO 2006025865	ICM	C22C
	IPCI	C22C0019-05 [I,C]; C22C0019-05 [I,A]
	ECLA	C22C019/05R

AB There is provided a method for depositing a modified **MCrAlY** coating on a turbine blade tip. The method utilizes laser deposition techniques to provide a metallurgical bond between a turbine blade substrate, such as a superalloy substrate, and the modified **MCrAlY** composition. Further the modified **MCrAlY** coating has sufficient thickness such that a post-welding grinding operation to size the turbine blade to a desired dimension will not remove the modified **MCrAlY** coating entirely. The modified **MCrAlY** coating thus remains on the finished turbine blade tip after grinding.

ST superalloy **MCrAlY** coating turbine blade laser deposition turbine

blade
 IT Turbines
 (blades, substrate; modified **MCrAlY** coatings on turbine blade tips with improved durability)
 IT Coating materials
 (heat-resistant; modified **MCrAlY** coatings on turbine blade tips with improved durability)
 IT Coating process
 (laser-induced; modified **MCrAlY** coatings on turbine blade tips with improved durability)
 IT Superalloys
 RL: PEP (Physical, engineering or chemical process); PYP (Physical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses)
 (substrate; modified **MCrAlY** coatings on turbine blade tips with improved durability)
 IT **202606-06-2, MCrAlY**
 RL: PEP (Physical, engineering or chemical process); PYP (Physical process); PROC (Process)
 (coating; modified **MCrAlY** coatings on turbine blade tips with improved durability)
 IT **880354-75-6 880354-76-7 880354-77-8 880354-78-9**
 RL: PEP (Physical, engineering or chemical process); PYP (Physical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses)
 (heat-resistant coating alloy; modified **MCrAlY** coatings on turbine blade tips with improved durability)
 IT **202606-06-2, MCrAlY**
 RL: PEP (Physical, engineering or chemical process); PYP (Physical process); PROC (Process)
 (coating; modified **MCrAlY** coatings on turbine blade tips with improved durability)
 RN 202606-06-2 HCAPLUS
 CN MCrAlY (9CI) (CA INDEX NAME)

*** STRUCTURE DIAGRAM IS NOT AVAILABLE ***

IT **880354-75-6 880354-76-7 880354-77-8 880354-78-9**
 RL: PEP (Physical, engineering or chemical process); PYP (Physical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses)
 (heat-resistant coating alloy; modified **MCrAlY** coatings on turbine blade tips with improved durability)
 RN 880354-75-6 HCAPLUS
 CN Nickel alloy, base, Ni 0-36, Pt 20-35, Cr 15-25, Co 15-22, Al 8-15, Hf 1-5, Re 1-5, Ru 1-5, Si 1-5, Ta 1-5, Zr 1-3, Y 0.1-1 (9CI) (CA INDEX NAME)

Component	Component Percent	Component Registry Number
Ni	0 - 36	7440-02-0
Pt	20 - 35	7440-06-4
Cr	15 - 25	7440-47-3
Co	15 - 22	7440-48-4
Al	8 - 15	7429-90-5
Hf	1 - 5	7440-58-6
Re	1 - 5	7440-15-5
Ru	1 - 5	7440-18-8
Si	1 - 5	7440-21-3

Ta	1	-	5	7440-25-7
Zr	1	-	3	7440-67-7
Y	0.1	-	1	7440-65-5

RN 880354-76-7 HCAPLUS

CN Nickel alloy, base, Ni 0-62, Pt 0-35, Cr 15-25, Co 15-22, Al 8-15, Hf 0-5, Re 0-5, Ru 0-5, Si 0-5, Ta 0-5, Zr 0-3, Y 0.1-1 (9CI) (CA INDEX NAME)

Component	Component Percent	Component Registry Number
Ni	0 - 62	7440-02-0
Pt	0 - 35	7440-06-4
Cr	15 - 25	7440-47-3
Co	15 - 22	7440-48-4
Al	8 - 15	7429-90-5
Hf	0 - 5	7440-58-6
Re	0 - 5	7440-15-5
Ru	0 - 5	7440-18-8
Si	0 - 5	7440-21-3
Ta	0 - 5	7440-25-7
Zr	0 - 3	7440-67-7
Y	0.1 - 1	7440-65-5

RN 880354-77-8 HCAPLUS

CN Nickel alloy, base, Ni 36, Cr 25, Co 20, Al 13, Re 3, Hf 2, Si 0.6, Y 0.3 (9CI) (CA INDEX NAME)

Component	Component Percent	Component Registry Number
Ni	36	7440-02-0
Cr	25	7440-47-3
Co	20	7440-48-4
Al	13	7429-90-5
Re	3	7440-15-5
Hf	2	7440-58-6
Si	0.6	7440-21-3
Y	0.3	7440-65-5

RN 880354-78-9 HCAPLUS

CN Nickel alloy, base, Ni 38, Cr 22, Co 20, Al 13, Re 3, Hf 2, Ru 1.5, Si 0.6, Y 0.3 (9CI) (CA INDEX NAME)

Component	Component Percent	Component Registry Number
Ni	38	7440-02-0
Cr	22	7440-47-3
Co	20	7440-48-4
Al	13	7429-90-5
Re	3	7440-15-5
Hf	2	7440-58-6
Ru	1.5	7440-18-8
Si	0.6	7440-21-3
Y	0.3	7440-65-5

L46 ANSWER 2 OF 2 HCAPLUS COPYRIGHT 2006 ACS on STN

AN 2003:470353 HCAPLUS

DN 139:40128
 ED Entered STN: 20 Jun 2003
 TI High-temperature corrosion-resistant alloy and ceramic thermal barrier coating material with metal bonding layer used in gas turbine
 IN Hidetaka, Oguma; Ikuo, Okada; Taiji, Torigoe; Kouji, Takahashi
 PA Mitsubishi Heavy Industries, Ltd., Japan
 SO Eur. Pat. Appl., 17 pp.
 CODEN: EPXXDW
 DT Patent
 LA English
 IC ICM C22C0019-05
 ICS C23C0004-00; C23C0030-00; C23C0028-00
 CC 56-3 (Nonferrous Metals and Alloys)
 Section cross-reference(s): 57
 FAN.CNT 1

	PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
PI	EP 1319730	A1	20030618	EP 2002-27556	20021209
	R: AT, BE, CH, DE, DK, ES, FR, GB, GR, IT, LI, LU, NL, SE, MC, PT, IE, SI, LT, LV, FI, RO, MK, CY, AL, TR, BG, CZ, EE, SK				
	JP 2003183752	A2	20030703	JP 2001-383689	20011217
	CA 2413649	AA	20030617	CA 2002-2413649	20021205
	US 2003148140	A1	20030807	US 2002-316070	20021211
	US 6756131	B2	20040629		
	CN 1427085	A	20030702	CN 2002-157165	20021217
PRAI	JP 2001-383689	A	20011217		

CLASS

PATENT NO.	CLASS	PATENT FAMILY CLASSIFICATION CODES
EP 1319730	ICM	C22C0019-05
	ICS	C23C0004-00; C23C0030-00; C23C0028-00
	IPCI	C22C0019-05 [ICM,7]; C23C0004-00 [ICS,7]; C23C0030-00 [ICS,7]; C23C0028-00 [ICS,7]
	IPCR	C22C0019-05 [I,A]; C22C0019-05 [I,C*]; C23C0004-02 [I,A]; C23C0004-02 [I,C*]; C23C0004-08 [I,A]; C23C0004-08 [I,C*]; C23C0028-00 [I,A]; C23C0028-00 [I,C*]; C23C0030-00 [I,A]; C23C0030-00 [I,C*]
	ECLA	C22C019/05R; C23C004/02; C23C004/08B; C23C028/00; C23C030/00
JP 2003183752	IPCI	C22C0019-05 [ICM,7]; C23C0004-08 [ICS,7]; C23C0004-10 [ICS,7]; C23C0014-14 [ICS,7]; F01D0005-28 [ICS,7]
	IPCR	C22C0019-05 [I,A]; C22C0019-05 [I,C*]; C23C0004-02 [I,A]; C23C0004-02 [I,C*]; C23C0004-08 [I,A]; C23C0004-08 [I,C*]; C23C0028-00 [I,A]; C23C0028-00 [I,C*]; C23C0030-00 [I,A]; C23C0030-00 [I,C*]
CA 2413649	IPCI	C22C0030-00 [ICM,7]; C09K0015-02 [ICS,7]; C09K0015-00 [ICS,7,C*]; C23C0024-04 [ICS,7]; C23C0024-00 [ICS,7,C*]; F01D0005-14 [ICS,7]; F01D0005-28 [ICS,7]
	IPCR	C22C0019-05 [I,A]; C22C0019-05 [I,C*]; C23C0004-02 [I,A]; C23C0004-02 [I,C*]; C23C0004-08 [I,A]; C23C0004-08 [I,C*]; C23C0028-00 [I,A]; C23C0028-00 [I,C*]; C23C0030-00 [I,A]; C23C0030-00 [I,C*]
US 2003148140	IPCI	B32B0015-04 [ICM,7]
	IPCR	C22C0019-05 [I,A]; C22C0019-05 [I,C*]; C23C0004-02 [I,A]; C23C0004-02 [I,C*]; C23C0004-08 [I,A]; C23C0004-08 [I,C*]; C23C0028-00 [I,A]; C23C0028-00 [I,C*]; C23C0030-00 [I,A]; C23C0030-00 [I,C*]
	NCL	428/629.000; 416/241.000B; 416/241.000R; 420/444.000; 428/633.000; 428/680.000
	ECLA	C22C019/05R; C23C004/02; C23C004/08B; C23C028/00;

C23C030/00
 CN 1427085 IPCI C22C0019-05 [ICM,7]; C23C0004-10 [ICS,7]; F01D0005-28 [ICS,7]
 IPCR C22C0019-05 [I,A]; C22C0019-05 [I,C*]; C23C0004-02 [I,A]; C23C0004-02 [I,C*]; C23C0004-08 [I,A]; C23C0004-08 [I,C*]; C23C0028-00 [I,A]; C23C0028-00 [I,C*]; C23C0030-00 [I,A]; C23C0030-00 [I,C*]
 AB A high-temperature corrosion-resistant alloy contains Co 0.1-12, Cr 10-30, Al 4-15, Y 0.1-5, Re 0.5-10 weight% and Ni in the balance, for example, Co 10, Cr 20, Al 6, Y 0.3, Re 4, and Ni in the balance. Optionally, the alloy may contain also Hf 0.01-0.7 and Si 0.01-1.5 weight%. The high temperature corrosion resistant alloy composition has an excellent oxidation resistance and ductility and is suitable for use in a bonding layer of a thermal barrier coating material. A thermal barrier coating material comprises a heat-resistant alloy base material (e.g., a superalloy containing Ni-22Cr-9Mo-8Co-1Al); a metal bonding layer disposed on said base material, and a ceramic layer disposed on said metal bonding layer. Said metal bonding layer is made into a film by using an electron beam deposition method.
 ST nickel alloy ceramic thermal barrier coating gas turbine
 IT Thermal barrier coatings
 (ceramic; high-temperature corrosion-resistant alloy and ceramic thermal barrier coating material with metal bonding layer used in gas turbine)
 IT Vapor deposition process
 (electron-beam; high-temperature corrosion-resistant alloy and ceramic thermal barrier coating material with metal bonding layer used in gas turbine)
 IT Turbines
 (high-temperature corrosion-resistant alloy and ceramic thermal barrier coating material with metal bonding layer used in gas turbine)
 IT Ceramic coatings
 (thermally insulating; high-temperature corrosion-resistant alloy and ceramic thermal barrier coating material with metal bonding layer used in gas turbine)
 IT 1344-28-1, Alumina, uses
 RL: TEM (Technical or engineered material use); USES (Uses)
 (component of thermal barrier coating; high-temperature corrosion-resistant alloy and ceramic thermal barrier coating material with metal bonding layer used in gas turbine)
 IT 148377-84-8 540788-28-1 540788-29-2 540788-30-5 540788-31-6
 540788-32-7 540788-33-8 540788-34-9 540788-35-0
 RL: CPS (Chemical process); PEP (Physical, engineering or chemical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses)
 (oxidation resistant and ductile alloy, bonding layer under ceramic thermal barrier coating; high-temperature corrosion-resistant alloy and ceramic thermal barrier coating material with metal bonding layer used in gas turbine)
 IT 540788-27-0
 RL: TEM (Technical or engineered material use); USES (Uses)
 (substrate, gas turbine superalloy; high-temperature corrosion-resistant alloy and ceramic thermal barrier coating material with metal bonding layer used in gas turbine)
 RE.CNT 6 THERE ARE 6 CITED REFERENCES AVAILABLE FOR THIS RECORD
 RE
 (1) Antolotti, N; WO 9902745 A 1999 HCAPLUS
 (2) Czech Norbert; US 5939204 A 1999 HCAPLUS
 (3) Siemens Ag; EP 0412397 A 1991 HCAPLUS
 (4) Siemens Ag; WO 9612049 A 1996 HCAPLUS
 (5) Sulzer Metco Us Inc; WO 0172455 A 2001 HCAPLUS

(6) Toennes, C; WO 9923265 A 1999 HCAPLUS

IT 540788-32-7 540788-35-0

RL: CPS (Chemical process); PEP (Physical, engineering or chemical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses)

(oxidation resistant and ductile alloy, bonding layer under ceramic thermal barrier coating; high-temperature corrosion-resistant alloy and ceramic thermal barrier coating material with metal bonding layer used in gas turbine)

RN 540788-32-7 HCAPLUS

CN Nickel alloy, base, Ni 62,Cr 18,Co 10,Al 6.5,Re 2,Si 1,Hf 0.5,Y 0.3 (9CI)
(CA INDEX NAME)

Component	Component Percent	Component Registry Number
Ni	62	7440-02-0
Cr	18	7440-47-3
Co	10	7440-48-4
Al	6.5	7429-90-5
Re	2	7440-15-5
Si	1	7440-21-3
Hf	0.5	7440-58-6
Y	0.3	7440-65-5

RN 540788-35-0 HCAPLUS

CN Nickel alloy, base, Ni 61,Cr 18,Co 10,Al 6,Re 2,Si 1.6,Y 1,Hf 0.8 (9CI)
(CA INDEX NAME)

Component	Component Percent	Component Registry Number
Ni	61	7440-02-0
Cr	18	7440-47-3
Co	10	7440-48-4
Al	6	7429-90-5
Re	2	7440-15-5
Si	1.6	7440-21-3
Y	1	7440-65-5
Hf	0.8	7440-58-6

=> fil reg

FILE 'REGISTRY' ENTERED AT 16:05:53 ON 27 SEP 2006

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DICTIONARY FILE UPDATES: 26 SEP 2006 HIGHEST RN 908803-03-2

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<http://www.cas.org/ONLINE/UG/regprops.html>

=> d l13 ide can

L13 ANSWER 1 OF 1 REGISTRY COPYRIGHT 2006 ACS on STN
RN 202606-06-2 REGISTRY
ED Entered STN: 12 Mar 1998
CN MCrAlY (9CI) (CA INDEX NAME)
MF Unspecified
CI AYS, MAN
SR CA
LC STN Files: CA, CAPLUS, USPAT2, USPATFULL

*** STRUCTURE DIAGRAM IS NOT AVAILABLE ***

PROPERTY DATA AVAILABLE IN THE 'PROP' FORMAT

95 REFERENCES IN FILE CA (1907 TO DATE)
95 REFERENCES IN FILE CAPLUS (1907 TO DATE)

REFERENCE 1: 145:193696
REFERENCE 2: 145:150707
REFERENCE 3: 145:112010
REFERENCE 4: 145:12678
REFERENCE 5: 144:437685
REFERENCE 6: 144:417707
REFERENCE 7: 144:335551
REFERENCE 8: 144:195925
REFERENCE 9: 143:481160
REFERENCE 10: 143:425652

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(FILE 'HOME' ENTERED AT 15:39:22 ON 27 SEP 2006)
SET COST OFF

FILE 'HCAPLUS' ENTERED AT 15:39:37 ON 27 SEP 2006
L1 1 S (WO2005-US06833 OR US2004-792003#)/AP, PRN
E HU/AU
E HU Y/AU
L2 526 S E3
L3 9 S E18
E HU YI/AU

L4 318 S E3,E35
E HU YIP/AU
L5 59 S E5
E HU NAME/AU
L6 1 S E4
E YIPING/AU
L7 1 S E7
E HEHMAN/AU
L8 9 S E23
E HONEYWELL/PA,CS
L9 4196 S E3,E4 OR HONEYWELL?/PA,CS
L10 667 S MCRALY
SEL RN L1

FILE 'REGISTRY' ENTERED AT 15:42:26 ON 27 SEP 2006

L11 5 S E1-E5
E MCRALY/CN
L12 1 S E3
L13 1 S L11 AND L12

FILE 'HCAPLUS' ENTERED AT 15:43:00 ON 27 SEP 2006

L14 95 S L13
L15 667 S L10,L14
L16 4 S L15 AND L1-L9

FILE 'REGISTRY' ENTERED AT 15:43:44 ON 27 SEP 2006

L17 3 S (CO AND CR AND AL AND Y AND PT AND HF AND SI AND RE AND RU AN
L18 2 S L17 NOT C/ELS
L19 1 S L17 NOT L18
L20 50 S (CO AND CR AND AL AND Y AND HF AND SI AND RE AND NI)/ELS
L21 13 S (CO AND CR AND AL AND Y AND HF AND SI AND RE AND RU AND NI)/E
L22 1 S L21 AND 9/ELC.SUB
L23 10 S (CO AND CR AND AL AND Y AND HF AND SI AND ZR AND TA AND RE AN
L24 0 S L23 AND 11/ELC.SUB
L25 4 S L11 AND L17-L24
L26 2 S L25 AND L17
L27 1 S L25 AND L22
L28 7 S L23 NOT L17
L29 3 S L20 AND 9/ELC.SUB
L30 1 S L25 AND L29
L31 6 S L18,L22,L25,L26,L27,L29,L30
L32 2 S L31 NOT L25
L33 3 S L20 AND 8/ELC.SUB
L34 8 S L31,L32,L33
L35 4 S L34 NOT L25
L36 2 S L35 NOT TA/ELS
L37 4 S L34 NOT L35
L38 6 S L36,L37
L39 3 S L20 AND 8/ELC.SUB
L40 1 S L29 NOT TA/ELS
L41 6 S L18,L39,L40
L42 6 S L38,L41

FILE 'HCAOLD' ENTERED AT 16:04:32 ON 27 SEP 2006

L43 0 S L42

FILE 'HCAPLUS' ENTERED AT 16:04:34 ON 27 SEP 2006

L44 2 S L42
L45 1 S L44 AND L1-L10,L14,L15
L46 2 S L44,L45

FILE 'USPATFULL' ENTERED AT 16:05:01 ON 27 SEP 2006
L47 1 S L42

FILE 'REGISTRY' ENTERED AT 16:05:17 ON 27 SEP 2006

FILE 'USPATFULL' ENTERED AT 16:05:24 ON 27 SEP 2006

FILE 'HCAPLUS' ENTERED AT 16:05:37 ON 27 SEP 2006

FILE 'REGISTRY' ENTERED AT 16:05:53 ON 27 SEP 2006

=> d que 118

L17 3 SEA FILE=REGISTRY ABB=ON PLU=ON (CO AND CR AND AL AND Y AND
PT AND HF AND SI AND RE AND RU AND NI)/ELS
L18 2 SEA FILE=REGISTRY ABB=ON PLU=ON L17 NOT C/ELS

=> d que 139

L20 50 SEA FILE=REGISTRY ABB=ON PLU=ON (CO AND CR AND AL AND Y AND
HF AND SI AND RE AND NI)/ELS
L39 3 SEA FILE=REGISTRY ABB=ON PLU=ON L20 AND 8/ELC.SUB

=> d que 140

L20 50 SEA FILE=REGISTRY ABB=ON PLU=ON (CO AND CR AND AL AND Y AND
HF AND SI AND RE AND NI)/ELS
L29 3 SEA FILE=REGISTRY ABB=ON PLU=ON L20 AND 9/ELC.SUB
L40 1 SEA FILE=REGISTRY ABB=ON PLU=ON L29 NOT TA/ELS

=> d que 124

L23 10 SEA FILE=REGISTRY ABB=ON PLU=ON (CO AND CR AND AL AND Y AND
HF AND SI AND ZR AND TA AND RE AND RU AND NI)/ELS
L24 0 SEA FILE=REGISTRY ABB=ON PLU=ON L23 AND 11/ELC.SUB

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